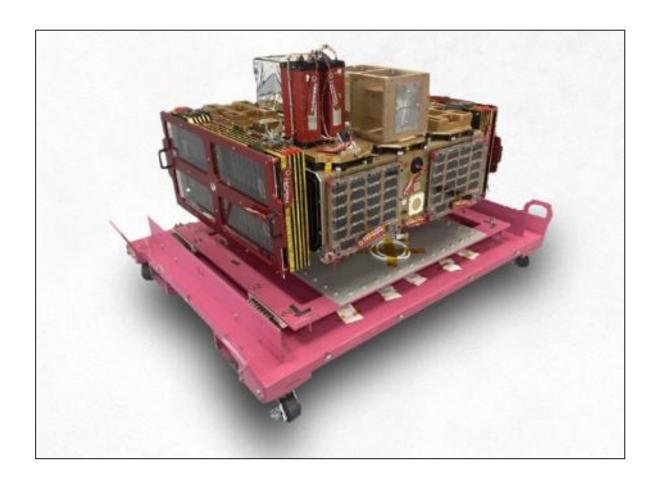


# IN-SPACE INFRASTRUCTURE SERVICES

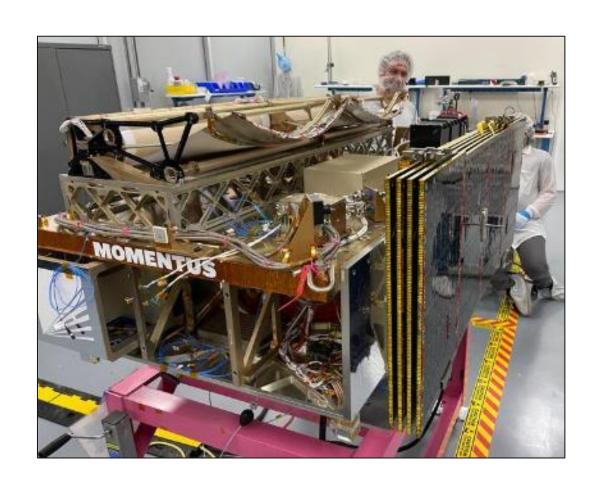
NOVEMBER 2025



#### **VEHICLES IN ORBIT**







#### Vigoride 3

 Deployed 8 Customer Satellites

#### Vigoride 5

- Hosted Caltech Space-Based Solar Power payload for 6month mission
- Deployed Cubesat
- Operated Microwave Electrothermal Thruster (MET)

#### Vigoride 6

- Delivered NASA & 4 Other Payloads
- Tape Spring Solar Array
   Demo

Three Launches in One Year

#### READY TO LAUNCH



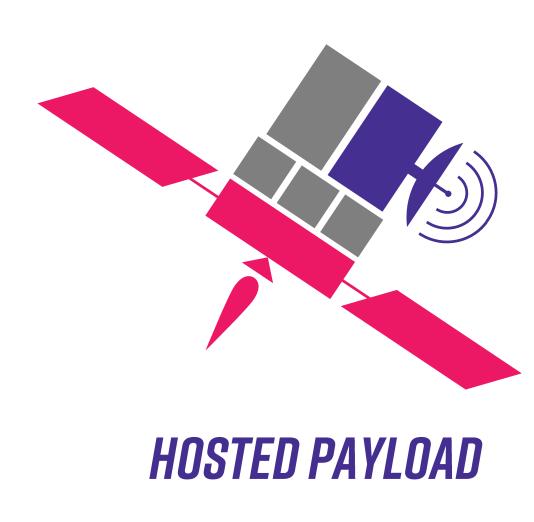
## Vigoride 7 with RPO Demo

- Completed environmental tests, ready to launch
  - RPO search and approach demo using low-cost sensors integrated
- DARPA NOM4D hosted payload
- In-space manufacturing demo
- Booked for Q1 2026 T16 launch
- Mission Fully Booked!

## AVAILABLE SERVICES

#### INFRASTRUCTURE SERVICES ENABLED BY ORBITAL SERVICE VEHICLES



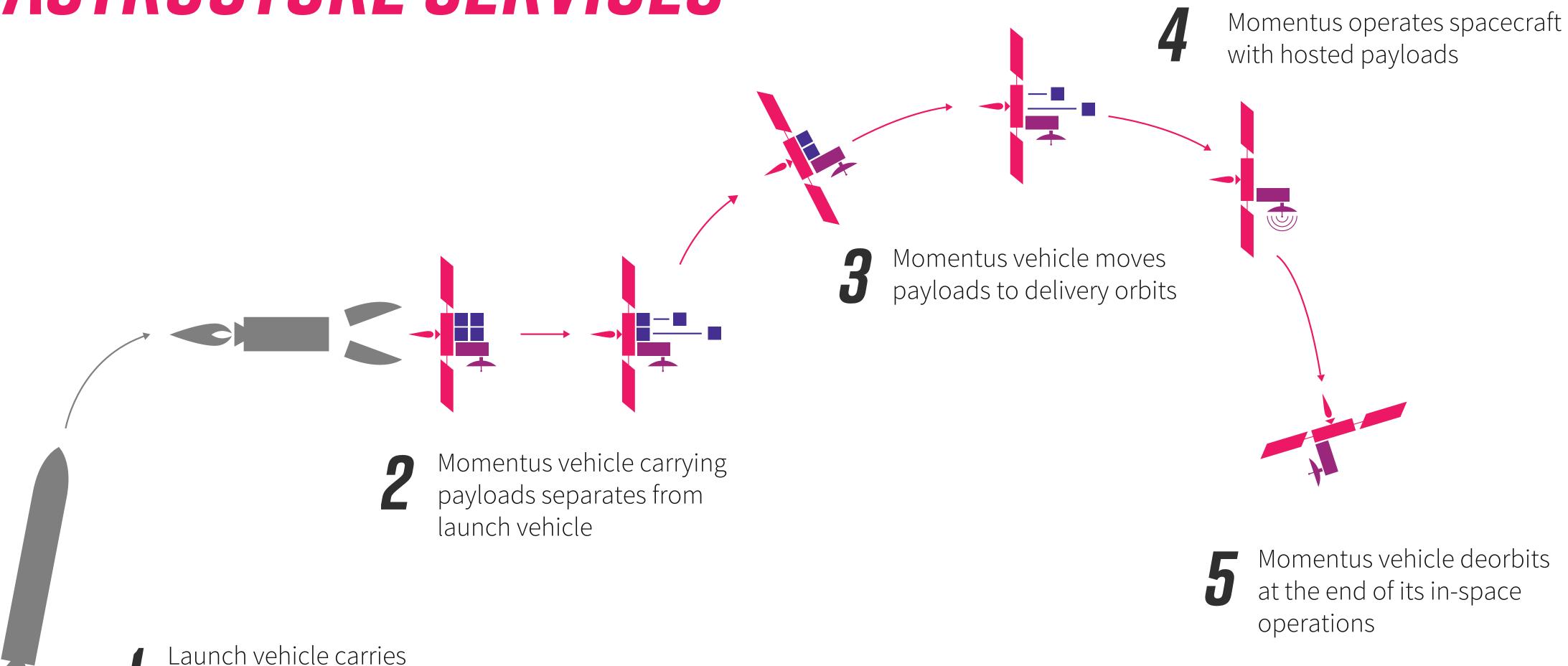




#### **COMING SOON**

REPAIR | UPGRADE | REFUELING | INSPECTION | REPOSITION | DEORBIT

## INFRASTRUCTURE SERVICES



Momentus vehicle with payloads to initial orbit

Mission Cost is Shared by Transportation Customers and Hosted Payloads

#### SERVICES

# LAUNGH & DELIVERY





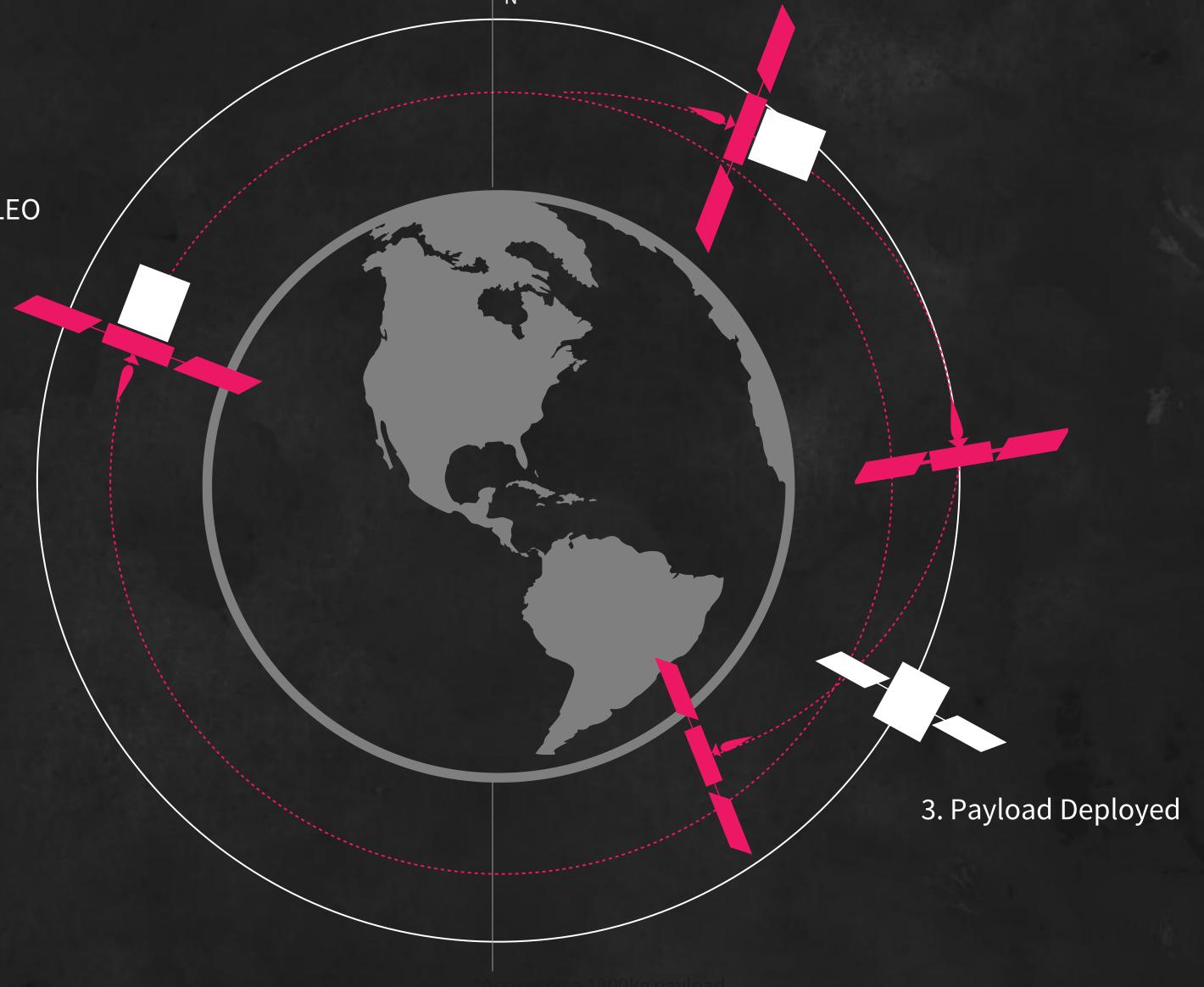
PRECISE INJECTIONS

2. Small Altitude +/- 20 km or/and +/-.1 degree Inclination Change

#### VIGORIDE OSV

| ORBITAL<br>PARAMETER | INJECTION ACCURACY |
|----------------------|--------------------|
| Semi-Major Axis      | +/- 2 km           |
| Inclination          | +/- 0.1 degrees    |
| RAAN                 | +/- 0.2 degrees    |

Launch To LEO
 Circular Orbit



## INGLINATION CHANGE

2. Adjustment in Inclination

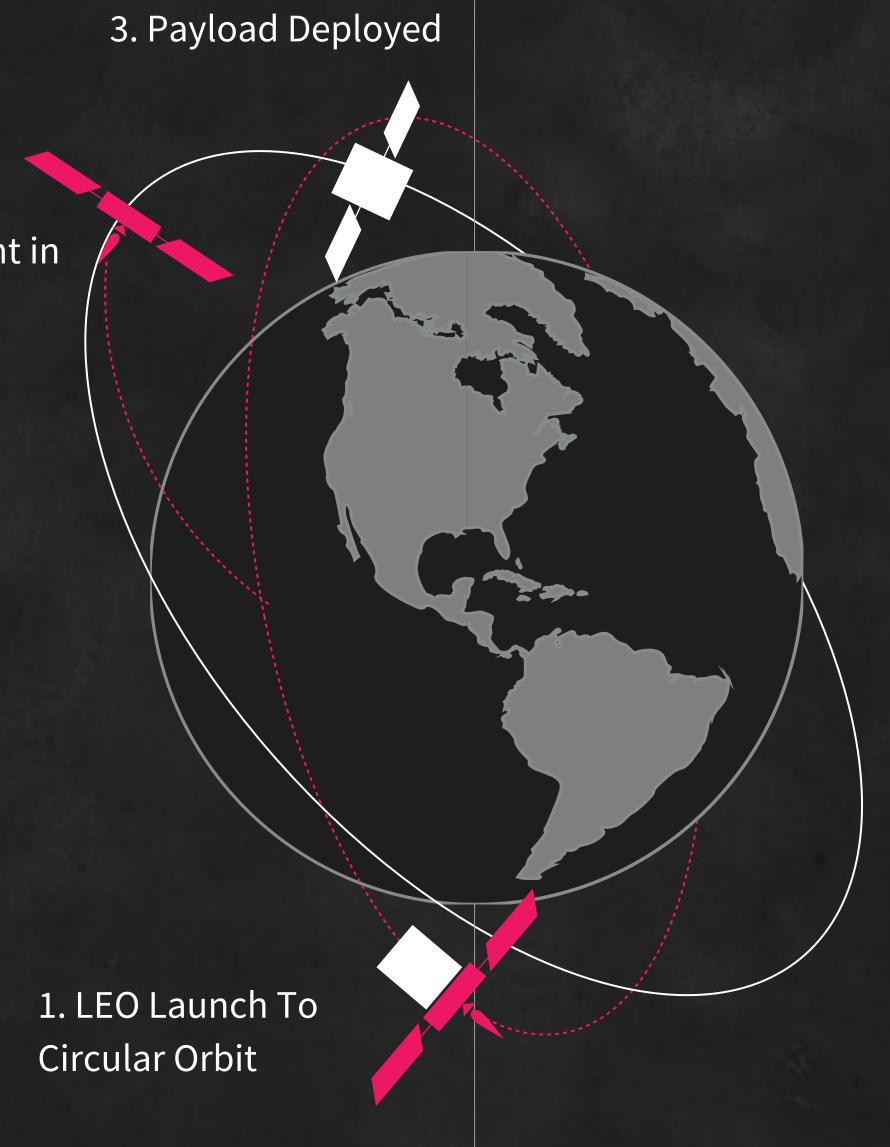
GEO

VIGORIDE OSV

Max Inclination Change (Degrees) Vigoride Configuration LEO MEO

MET (Water) 6.5 9.7 15.4

HET (Krypton) 17 26.9 44.4



Ν



## ALTITUDE CHANGE

VIGORIDE OSV

Vigoride Circular Orbit Altitude Range From (km)

Configuration

LEO

GEO

MET (Water)

Re-entry to 2,265

5,880 to 18,000

MEO

19,700 to 73,700

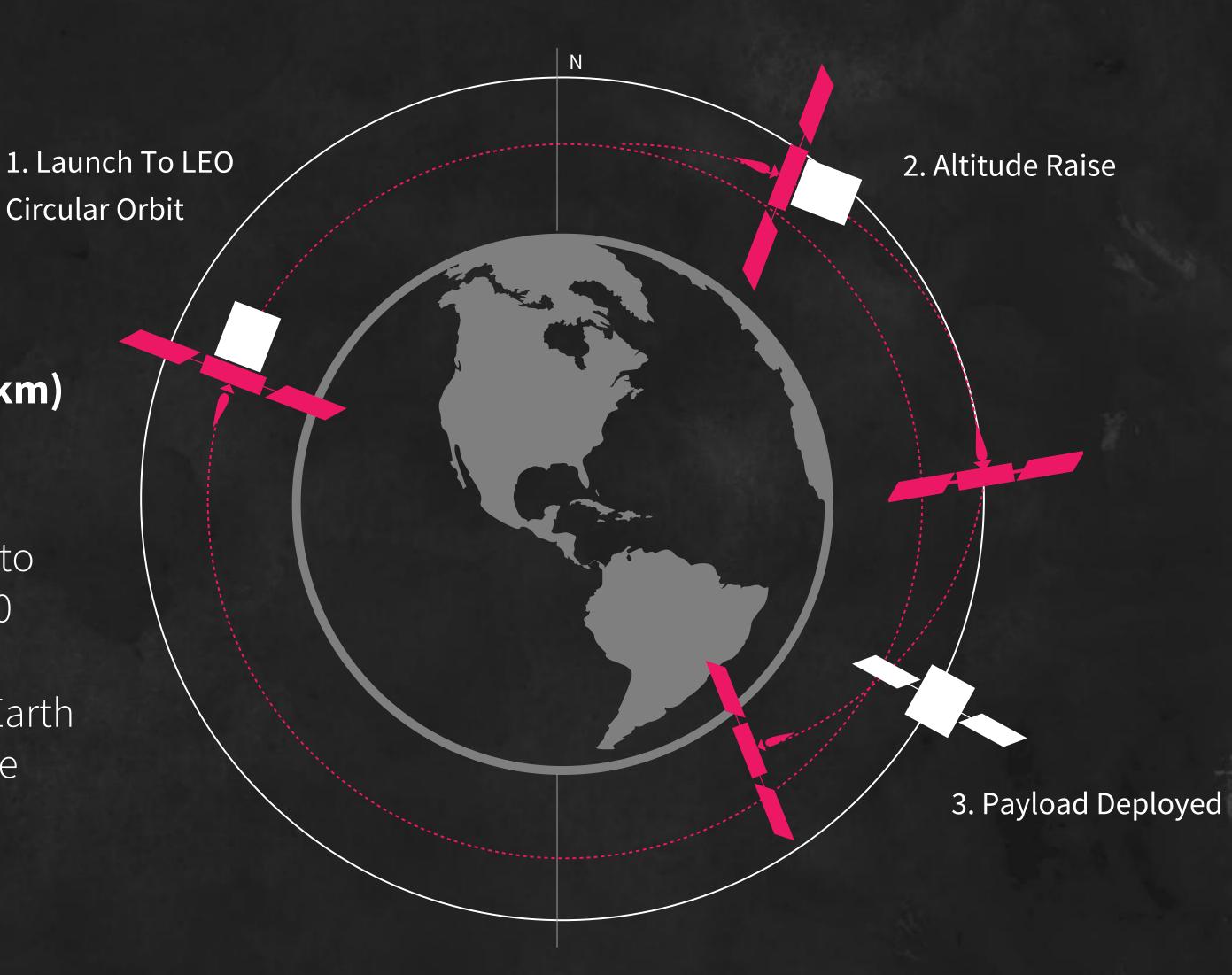
HET (Krypton)

Re-entry to 17,353

Re-entry to 209,293

2,800 to Earth

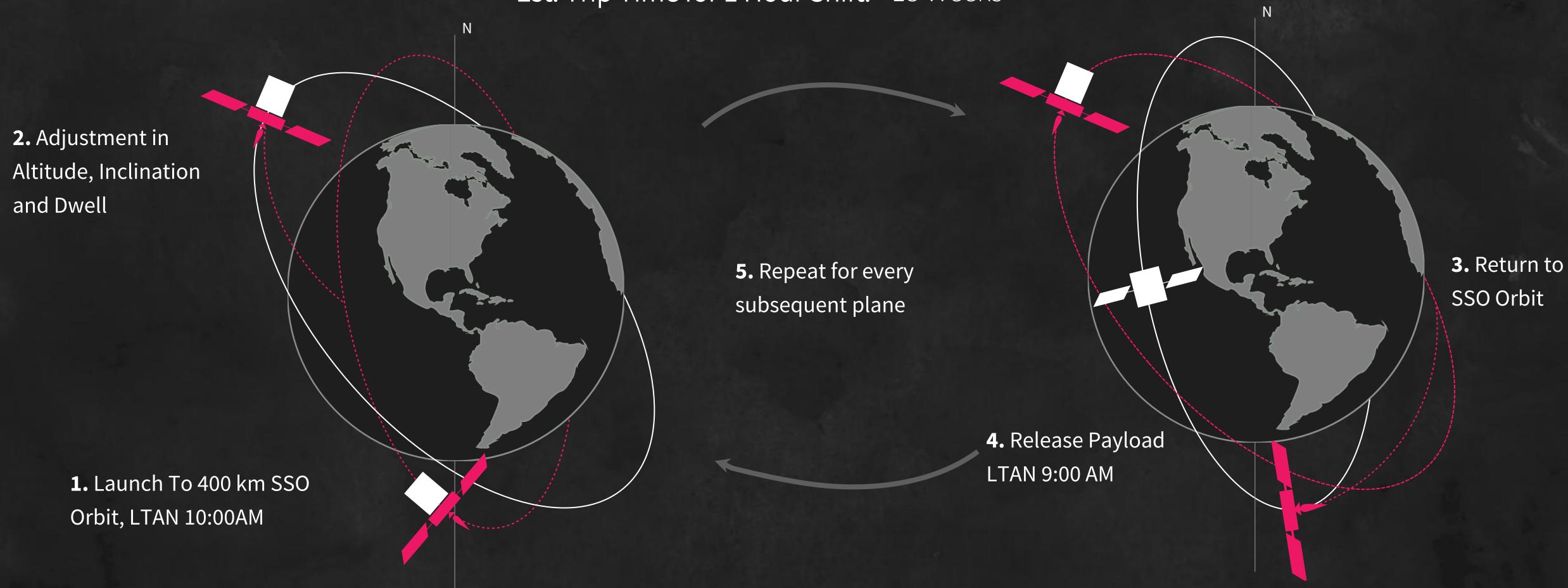
Escape



## LTAN SHIFT

#### VIGORIDE OSV

Est. Trip Time for 1 Hour Shift: <13 Weeks



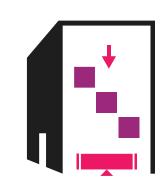


## ILLUSTRATIVE ORDER OF OPERATIONS

Momentus' capabilities make access to space significantly more affordable by combining rideshare launch with low-cost last mile delivery. Arriving in-space within the faring of nearly any rocket, our service vehicles carry a variety of customer satellites to very specific, custom orbits.



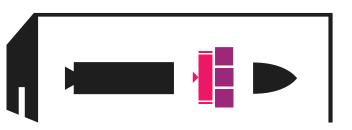
Customers ship payloads to Momentus as late as L-4 months



2 Momentus integrates payloads with vehicle



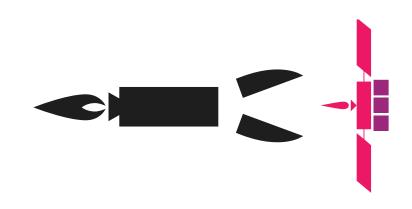
3 Momentus ships to launch service provider



Momentus vehicle integrates with launch vehicle (LV)



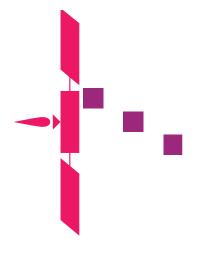
5 LV carries Momentus vehicle with payloads to initial orbit



6 Momentus vehicle carrying payloads separates from LV



Momentus vehicle moves payloads to delivery orbits



8 Momentus vehicle delivers payloads to custom, final orbits

SERVICES

## HOSTED PAYLOAD



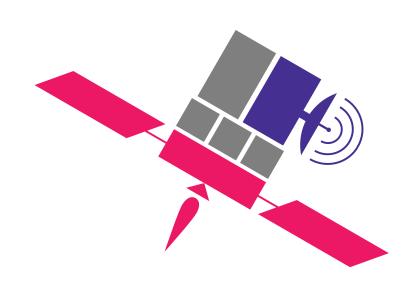


### HOSTED PAYLOAD

Hosted payload services will include:

- Payload & Launch Vehicle Integration
- Licensing
- Launch & In-Space Transportation
- Operation of the Momentus Vehicle
- Power
- Data
- Pointing
- Communication

Our hosted payload services offer turn-key satellite bus capabilities and allow customers to focus on what's important – the **payload** and the **mission**.



#### RIDESHARE SERVICE

Shared rides with fellow customer spacecraft for economical access to custom orbit destinations. Co-manifested customers could include both hosted and deployed payloads.



#### DEDICATED SERVICE

Exclusive use of a Momentus vehicle platform for your mission, accommodating larger hosted payload mass, volume, and power requirements. The entire in-space transportation capability of the Momentus vehicle is dedicated to your hosted payload mission.



## HOSTED PAYLOAD: AVAILABLE VOLUME

Momentus' hosted payload service includes a significant amount of available volume for hosted payloads. This volume can be dedicated to larger hosted payloads or customizable configurations of numerous rideshare deployable and hosted payloads.

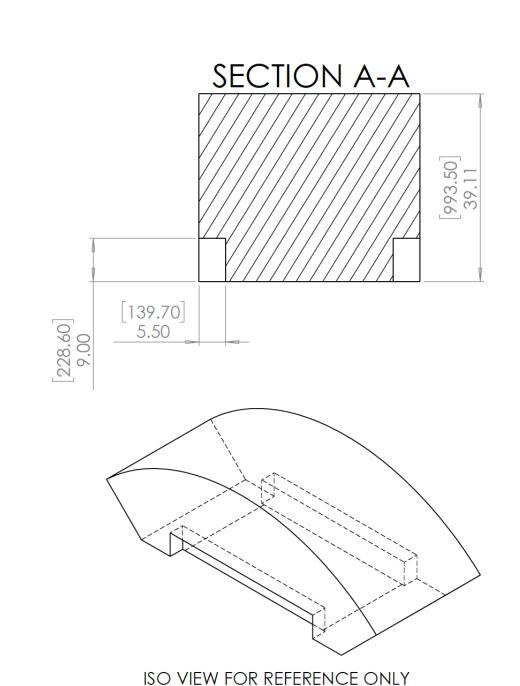
For alternative volume configurations please consult a Momentus Sales Representative (sales@momentus.space)

### 2246.82 3030.86 119.33 PAYLOAD DECK AREA SHADED [609.60] $\emptyset$ 24.00 CG ALLOWABLE 1549.40 61.00 1600.20

NOTE: DIMENSIONS IN [XXX.XX] ARE PROVIDED IN mm FOR REFERENCE ONLY.

VIGORIDE ENVELOPE FOR HOSTED PAYLOADS

(FALCON 9 XL PLATE LAUNCH CONFIGURATION)





#### VIGORIDE HOSTED PAYLOAD CAPABILITY

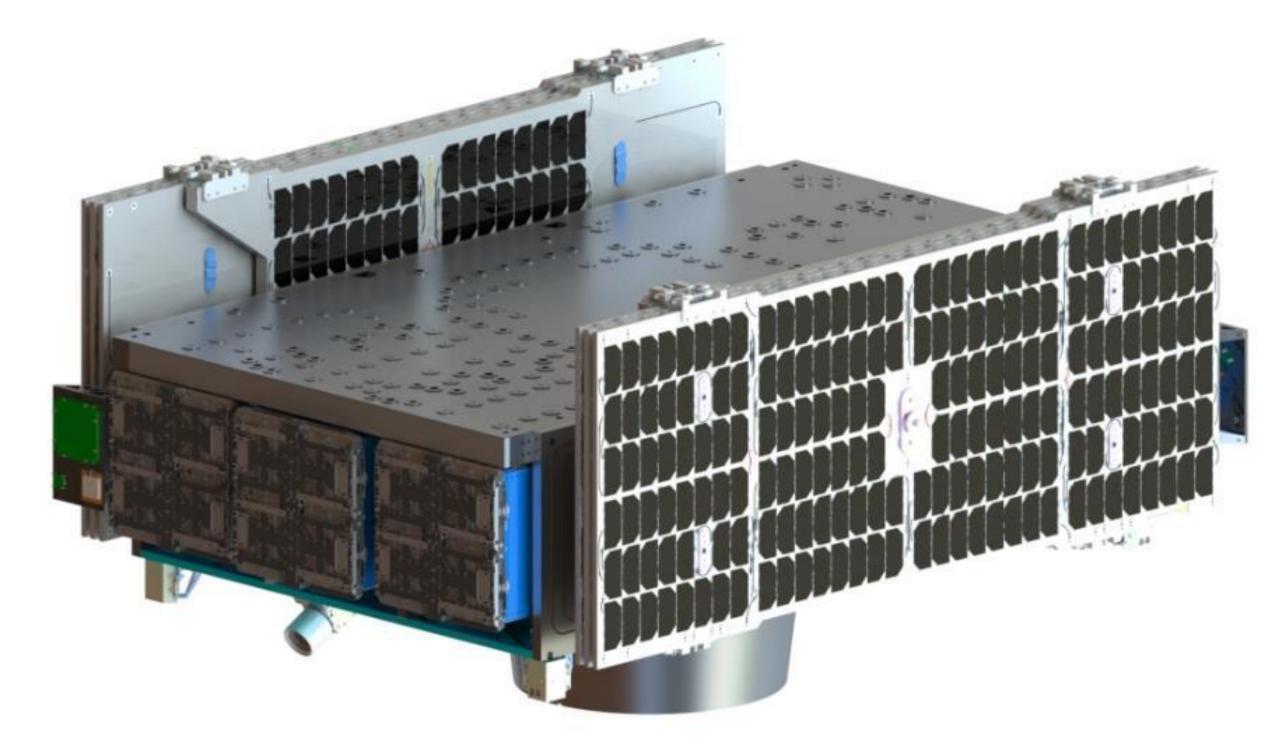
Bus designed to be rapidly configured for a wide range of mission requirements

Payloads up to ~800 kg

**Deploy up 72U of Cubesats** 

Transport and operate at custom inclinations and/or altitudes as required (currentle LEO) up to 1km/s Δv

Payload deck 1.5m x 0.9m



Pointing Stability tunable to 10 arc-seconds

Product roadmap includes radiation hardening for GEO/Lunar environment

~1 kw on-orbit average power; ~3 kw peak Pumped fluid loop cooling

Dual redundant systems for assured reliability

Baseline S/X-Band TT&C
NSA Type 1 Certified
Encryption Option
(Innoflight KI-103)

TRL 9 Water Propellant based Microwave Electrothermal Thrusters and reaction control

Modular propulsion: electric, chemical, or MET



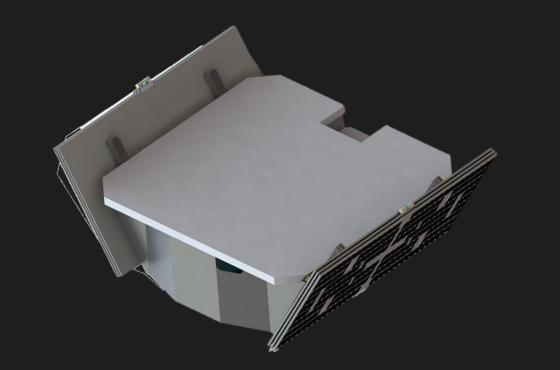
## VIGORIDE ORBITAL SERVICE VEHICLES

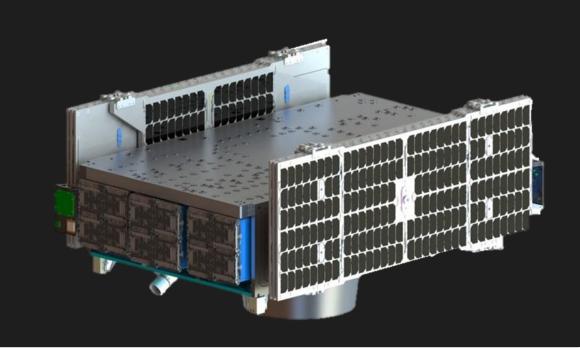




## MOMENTUS BUS PERFORMANCE







| Parameter                         | Vigoride Lite  | Vigoride Bus                                   | Vigoride OSV              |
|-----------------------------------|--|--|---------------------------|
| Launch Interface                  | ESPA / SX Full Plate   | ESPA Grande / SDA                              | ESPA Grande / SX XL Plate |
| Launch Mass                       | 200 kg   | 550 kg (Side-mount), 1000 kg (Vertical-mount)  |                           |
| Payload Mass                      | 75 kg  | 280 kg (Side-mount) or 800 kg (Vertical-mount) |                           |
| Payload Mounting Area             | 4900 cm <sup>2</sup>   | 27,832 cm <sup>2</sup>                         | 20,645 cm <sup>2</sup>    |
| Underdeck Cubesat Capacity        | -  | -  | 72U                       |
| Payload Peak Power                | 750 W  | 3 kW   |                           |
| Payload Orbit Averaged Power      | 200 W  | 1 kW   |                           |
| Payload Thermal Dissipation @ 50C | 100 W  | 500 W  | 300 W                     |
| Voltage                           | 40 V to 60 V Unregulated, 28 V Regulated (low- and high-power) |  |                           |
| Base Model Pointing Control       | 50 mdeg  |  | 1 degree                  |



## MOMENTUS PRODUCT LINE PERFORMANCE

| Parameter                          | Base Performance                      | Enhanced Performance                | Premium Performance |  |
|------------------------------------|---------------------------------------|-------------------------------------|---------------------|--|
| Mission Data Downlink Rate         | 50 kbps (TC&R)                        | 300 Mbps (High gain RF)             | 1 Gbps (Optical)    |  |
| Data Interfaces                    | GigE, RS-485, RS-422, USB             |                                     |                     |  |
| Data Storage                       | 160MiB Vigoride /<br>256 GiB M-series | 4 TB                                | 12 TB               |  |
| Positional Knowledge<br>(Absolute) |                                       | 500 m                               |                     |  |
| Positional Knowledge<br>(Relative) | 500 m                                 | 500 m 5% when < 100 m (RPO Package) |                     |  |
| Pointing Control                   | *See prior chart                      | 50 mRad/100 arcsec                  | 50 uRad/10 arcsec   |  |
| Slew Rate                          | 1.5 deg/sec                           |                                     |                     |  |
| Thrust                             | 230 mN (MET)                          | 4N (Chemical)                       | 78 mN (HET)         |  |
| Delta-V                            | 1 km/s (MET)                          | 500 m/s (Chemical)                  | 2 km/s (HET)        |  |
| Ground Antenna Access              | Global: > 70 Locations                |                                     |                     |  |
| Orbital Domain                     | LEO (20 kRad)                         | GEO – CisLunar (100 kRad)           |                     |  |
| Orbital Lifetime                   | 5 Years                               | TBD, > 5 years                      |                     |  |



